


Article

# Distinct fracture patterns in construction steels for reinforced concrete – a review.

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**Abstract:** Steel is one of the most widely used materials in construction. Nucleation growth and coalescence theory is usually employed to explain the fracture process in ductile materials, such as many metals. The typical cup-cone fracture pattern has been extensively studied in the past, giving rise to numerical models able to reproduce this pattern. Nevertheless, some steels, such as the eutectoid steel used for manufacturing prestressing wires, does not show this specific shape but a flat surface with a dark region in the centre of the fracture area. Late studies have deepened the knowledge on these distinct fracture patterns, giving light on some aspects that help to understand how damage begins and propagates in each case. The numerical modelling of both fracture patterns have also been discussed and reproduced with different approaches. This work reviews the main lately advances in the knowledge on this subject, particularly focusing on the experimental work carried out by the authors.

**Keywords:** Steel; Fracture Mechanics; Tensile Test; Cohesive zone model; Internal damage; XRCT.

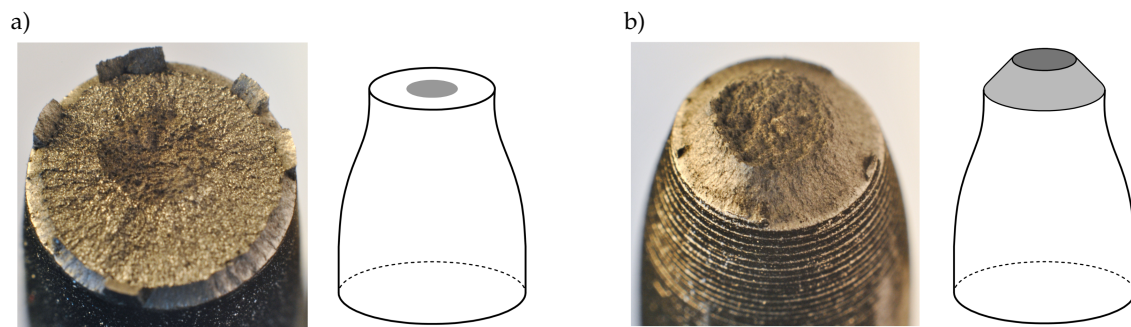
## 1. Introduction

Steel is, with concrete, the most extended material in construction and civil engineering works. Its strength and ductility make it of special interest when addressing structural safety issues, since it enables stress distribution with adjacent elements, allowing a higher amount of energy to dissipate before failure. However, some aspects related with its failure behaviour remain unclear, especially in those steels that do not present the classical cup-cone fracture surface, which has been extensively studied in the past.

The mechanical characterization of this type of materials is usually reduced to obtaining their elastic parameters, elastic modulus  $E$  and Poisson's ratio  $\nu$ . These values are generally obtained by means of a tensile test, which is standardised by EN ISO 6892 [1], and allows obtaining with precision the load-strain diagram up to the maximum loading point. Nevertheless, difficulties arise when the behaviour after the maximum load point needs to be defined, which usually leads to neglecting that information of the test. This final part of the load-strain diagram is, however, of great interest since it is directly related to the maximum energy that can be absorbed by an structural element before collapsing, which goes together with the structural safety and may help, for instance, to distinguish between accidental damage and induced damage.

29 Regarding the construction steels fracture mechanisms, the cup-cone fracture pattern, typical of  
 30 very ductile steels and shown in Figure 1b), is very well known and has been extensively studied in  
 31 the past [3–10]. The mechanisms involved in it are clearly identified: the central zone corresponds to  
 32 a process of nucleation, growth and coalescence of microvoids while the surrounding inclined lips,  
 33 usually referred to as shear lips [3,11], develop due to cleavage. Nevertheless, not all steels show the  
 34 same fracture behaviour, for instance, the pearlitic steel used for manufacturing prestressing wires  
 35 presents a flat fracture surface, perpendicular to the loading direction and with a circular dark area in  
 36 the center (see Figure 1a)). For the sake of clarity, in the rest of the text the flat pattern will be referred  
 37 to as type 1 and the cup-cone pattern as type 2.

38 Other authors have studied the steel used in prestressing steel wires, which shows the flat fracture  
 39 pattern called type 1, focusing their interest on the effect of cold-drawing on the developed fracture  
 40 mechanism [12–14].



**Figure 1.** Fracture surfaces on 9mm-diameter specimens of two steels with different fracture patterns after testing under tension: a) Fracture pattern type 1, b) Fracture pattern type 2 [15].

41 Despite the characteristics of steels have been studied for a long time, there are still some issues  
 42 that remain unclear, especially regarding the plastic behaviour after maximum loading and, more  
 43 precisely, the mechanisms that unchain final failure in steels exhibiting different fracture patterns.  
 44 The objective of this work is to review recent experimental and numerical advances dealing with  
 45 fracture mechanics in steels, with special emphasis on the type 1 pattern, flat fracture surface, since  
 46 the type 2, cup-cone pattern, is better known and has been widely studied in the past. Firstly, some  
 47 relevant experimental results that analyse both fracture patterns are presented and secondly, the main  
 48 numerical models used with metals are briefly overviewed, also describing a lately approach based on  
 49 Linear Elastic Fracture Mechanics (LEFM) considerations.

## 50 2. Experimental results on steel specimens under tensile loading

51 The following results correspond to two steels that are representative of both mentioned fracture  
 52 patterns. Table 1 shows their chemical composition, with Material 1 being an eutectoid steel used for  
 53 manufacturing prestressing wires before cold-drawing, which exhibits fracture pattern type 1 and  
 54 Material 2 a low-carbon steel ( $< 0.24\%C$ ), used as reinforcement in concrete structures, which shows  
 55 the fracture pattern denoted as type 2.

**Table 1.** Chemical composition of both materials in %.

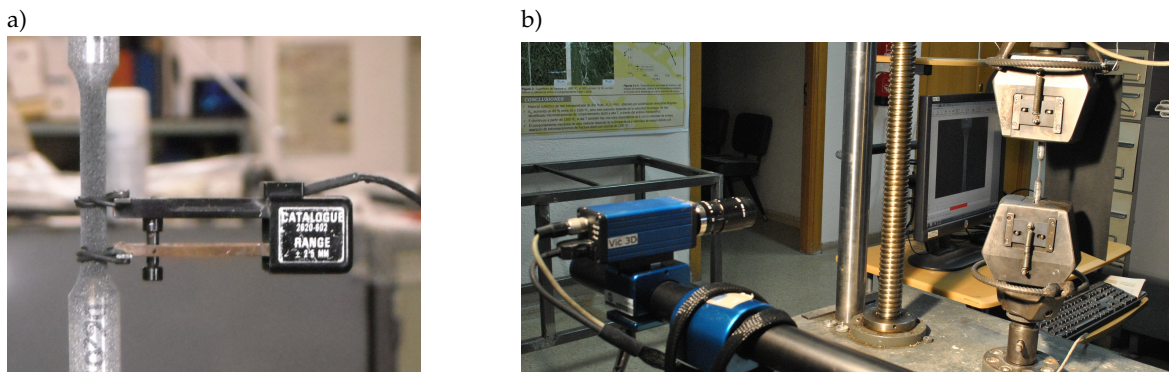
Mat.	C	Si	Mn	P	S	Cr	Mo	Ni	Cu	Al	Ti	Nb	V	N
1	0.83	0.25	0.72	0.012	0.004	0.24	<0.01	0.02	0.01	<0.003	<0.005	<0.005	<0.01	0.0097
2	0.22	0.18	1.00	0.024	0.042	0.08	0.03	0.14	0.46	<0.003	<0.005	<0.005	<0.01	0.0113

## 56 2.1. Stress-strain diagrams

57 When a steel bar is tested under tension there are several issues that may affect the results, such as  
58 the specimen length, the specimen radius, the initial gage length used or even the technique employed  
59 for measuring strain. For a complete description of the following results, the reader is referred to  
60 [15,16].

### 61 2.1.1. Strain measuring technique

62 As already mentioned, the last part of the stress-strain diagram is usually discarded because when  
63 the specimen is loaded beyond its maximum bearing capacity, strain is localised in a necking region  
64 which makes the analysis difficult. In fact, this is true if the classical strain measuring devices are used,  
65 such as conventional extensometers as that shown in Figure 2a), which only provide information of  
66 elongation between two specific points in the specimen and their results are only valid up to the  
67 maximum load point, since necking may take place out of the gauge length or, at least, not centered  
68 with the reference points used to measure strain. Nevertheless, in the last decades digital image  
69 correlation systems (DIC) have become widely extended, which overcome this limitation [17–22].  
70 These systems use one or two high-definition video cameras (depending on whether results are needed  
71 in 2D or 3D), which record a marked specimen throughout the test; Figure 2b) shows the experimental  
72 setup for a 2D case. The obtained pictures are postprocessed with a specific software that can keep  
73 track of the speckles marked on the specimen surface, providing relative displacements between any  
74 desired pair of points or even providing a strain mapping of the specimen surface. This has a major  
75 advantage for analysing the last part of the stress-strain diagram in a tensile test: the deformation can  
76 always be tracked between points that are equally spaced from the eventual fracture plane.

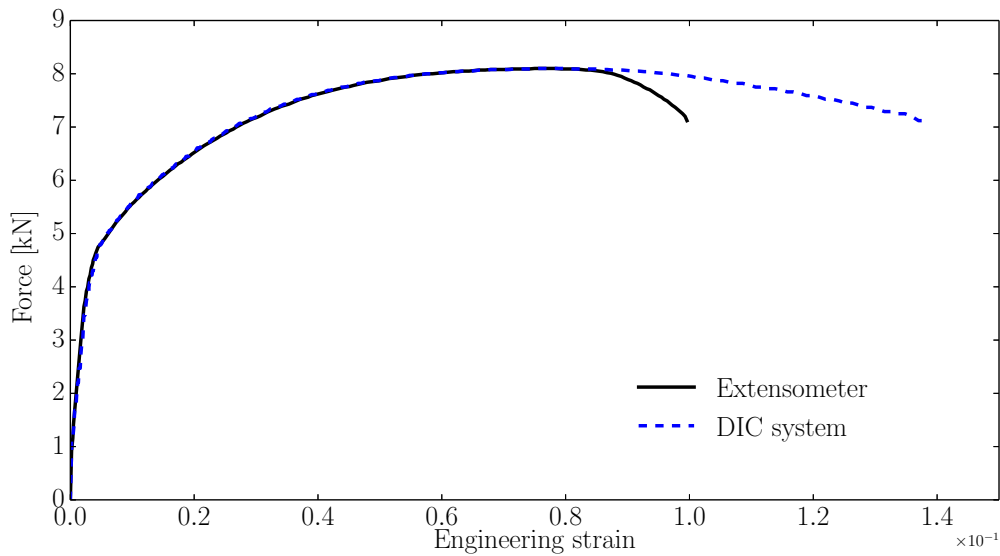


**Figure 2.** a) A conventional extensometer on a cylindrical specimen, b) Setup for measuring strains with a digital image correlation system [16]

77 Figure 3 shows the fracture diagram obtained for a tensile test where strain was monitored with  
78 both systems; DIC proves its capability for providing data up to the eventual fracture instant, while the  
79 conventional extensometer only provides valid information up to the maximum load point. Therefore,  
80 if DIC systems are used for monitoring strain, there is in principle no reason why the information  
81 obtained after peak should be neglected.

### 82 2.1.2. Influence of the specimen radius and the initial gauge length

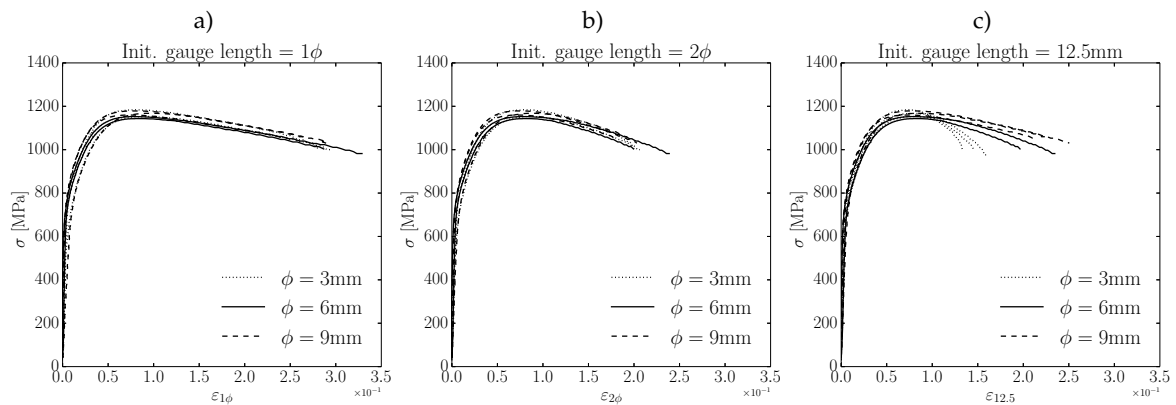
83 When testing a cylindrical steel bar under tension, several parameters must be defined, such as  
84 the initial gauge length and the diameter and length of the specimen. Figures 4 and 5 show several  
85 engineering stress-strain diagrams for Materials 1 and 2, respectively; all tests were monitored with  
86 DIC, so the strain was measured with a gauge length centered with the fracture cross-section and  
87 obtained up to the eventual failure of the specimen. Three specimen diameters were considered, 3 mm,



**Figure 3.** Force-engineering strain curves of a tensile test simultaneously obtained with a conventional extensometer and a DIC system [16]

88 6 mm and 9 mm, and three gauge lengths compared, one of them of a fixed length equal to 12.5 mm  
 89 and two proportional to the specimen diameter, once and twice the specimen diameter.

90 In both materials, when proportional-to-the-diameter gauge lengths are considered, the results  
 91 obtained are always similar, no matter the specimen diameter, but when a fixed gauge length is  
 92 employed, clear differences are found. When necking develops, the strain field is not uniform anymore  
 93 along the specimen and concentrates on very localised region of the bar; therefore, strains measured  
 94 with a fixed gauge length on specimens of different diameters provide a certain average strain that  
 cannot be compared among them. See section 2.2 for further details.

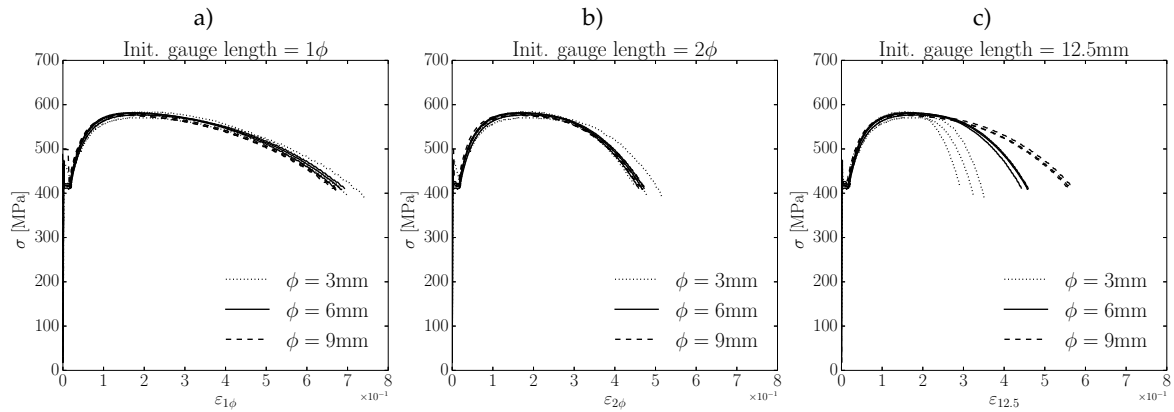


**Figure 4.**  $\sigma - \varepsilon$  curves obtained with Material 1 specimens of distinct radii and using several initial gauge lengths: a) initial gauge length= $1\phi$ , a) initial gauge length= $2\phi$ , a) initial gauge length=12.5 mm. [23]

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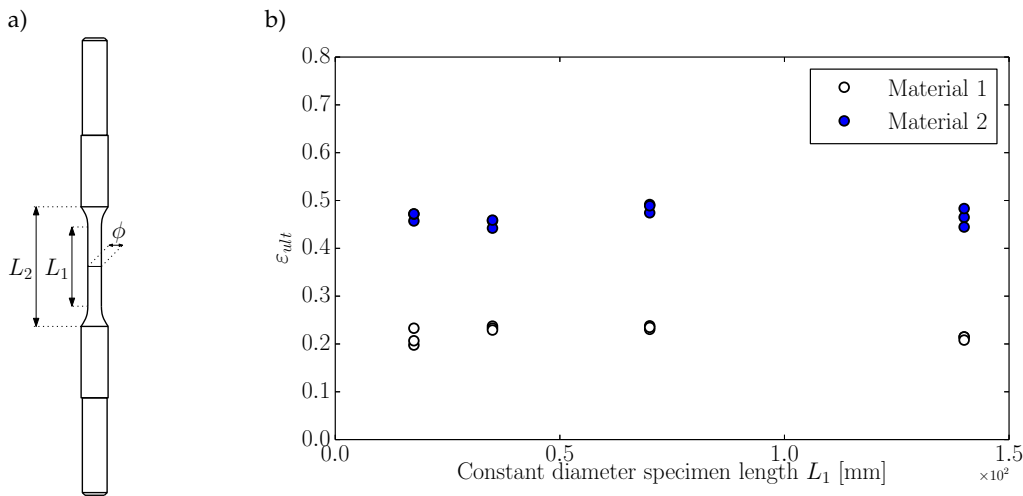
### 96 2.1.3. Influence of the specimen length

97 The length of the constant cross-section length in a specimen ( $L_1$  in Figure 6a)) is limited by the  
 98 standards to avoid too short specimens that cannot develop the whole necking process. In principle,  
 99 once the specimen dimensions meet the requirements defined in [1], this can be designed with any



**Figure 5.**  $\sigma - \varepsilon$  curves obtained with Material 2 specimens of distinct radii and using several initial gauge lengths: a) initial gauge length= $1\phi$ , a) initial gauge length= $2\phi$ , a) initial gauge length=12.5 mm. [23]

100 desired length. In [16] this issue was studied by testing 6mm-diameter specimens with increasing  
 101 values of  $L_1$ . In both materials, 1 and 2, the specimen length did not seem to affect the ultimate strain  
 102 as can be observed in Figure 6), that shows the results for distinct specimen lengths ( $L_1$ ) using a fixed  
 103 initial gauge length of 12.5 mm.



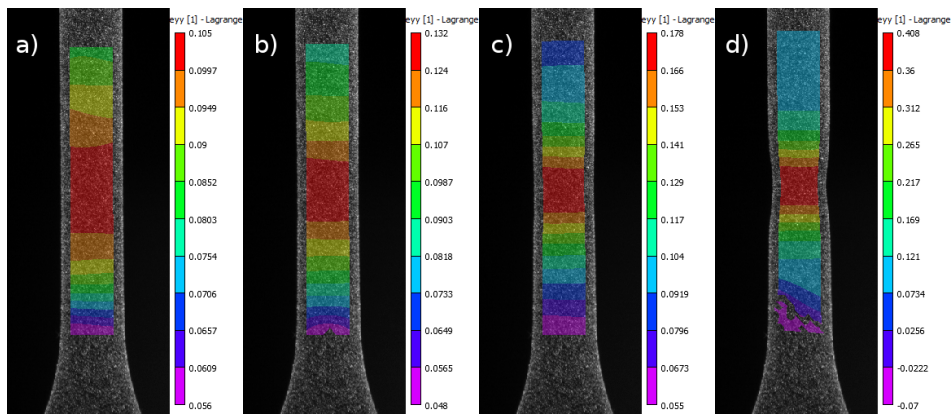
**Figure 6.** a) Specimen parameters for tensile tests, b) influence of the specimen length on the ultimate strain under tensile loading for an initial gauge length of 12.5 mm. [16].

## 104 2.2. Strain fields maps

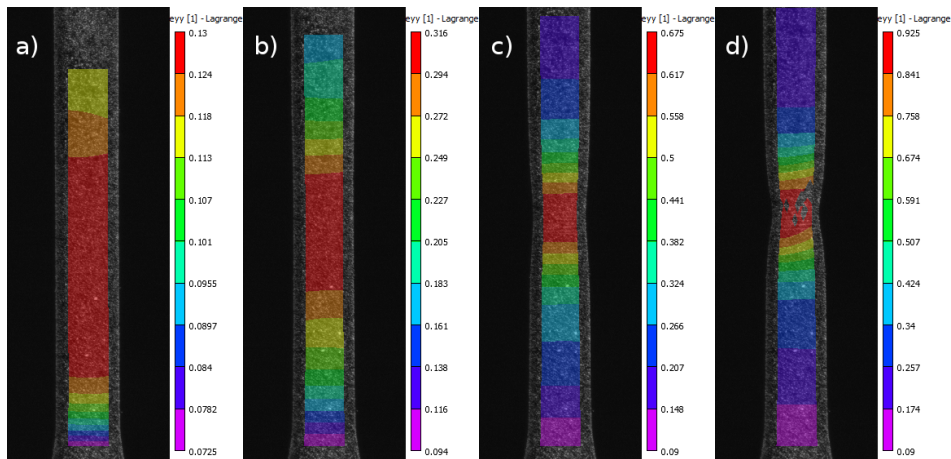
105 One of the reasons that make DIC a powerful technique is that, differently from the conventional  
 106 extensometers, which provide elongation values between only two points of the specimen, it can be  
 107 used to obtain the strain maps at the surface of the specimen. Figures 7 and 8 show the longitudinal  
 108 strain evolution on specimens of Materials 1 and 2, respectively; in both cases sub-figure a) represents  
 109 the strain map at the maximum load instant and d) at the very last instant before failure, with b) and c)  
 110 being intermediate images between a) and d).

111 The strain maps help to understand the differences observed when a fixed gauge length is used  
 112 with distinct diameter specimens. In Figures 4c) and 5c), the curves for all diameters were coincident up  
 113 to the maximum load point and bifurcated later, showing smaller values of strain for smaller diameters.

114 The reason is that, as Figures 7 and 8 show, the strain gradient becomes greater as the tensile test  
 115 progresses; therefore, if a proportional-to-the-diameter gauge length is used, the strain measured is  
 116 comparable among distinct diameter specimens, proving that the necking phenomenon is proportional  
 117 to the diameter, which explains why the curves in these cases are coincident beyond the maximum load  
 118 point. Nevertheless, when a fixed gauge length is used, the gauge provides an average strain value in  
 119 the part of the specimen that is between the selected points, therefore, in the case of smaller diameter  
 120 specimens, the gauge length includes parts of the specimen relatively distant from the necking region,  
 121 which results in smaller values of  $\epsilon$ , while in the case of larger diameter specimens the gauge length  
 122 includes regions that present comparatively higher strain gradients and, therefore, the average values  
 123 of  $\epsilon$  are larger. The strain gradient is higher in the case of Material 2 compared with the strain maps of  
 124 Material 1, which explains why the bifurcation is more evident in Figure 5 than in Figure 4.



**Figure 7.** Evolution of the vertical strain field from the maximum load point (a) to the instant just before failure (d) for a 3-mm diameter specimen of Material 1. [23]



**Figure 8.** Evolution of the vertical strain field from the maximum load point (a) to the instant just before failure (d) for a 3-mm diameter specimen of Material 2. [23]

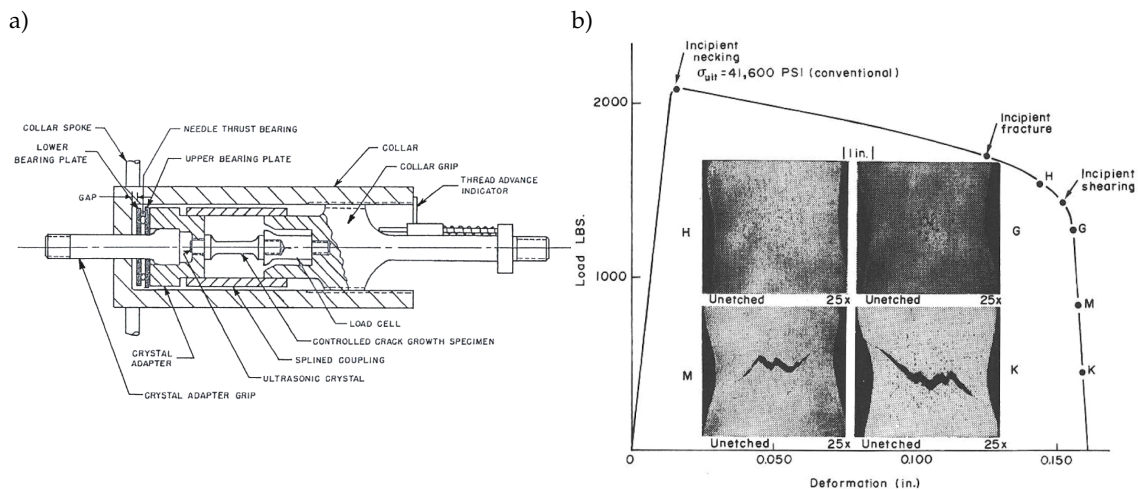
### 125 2.3. Analysis of the fracture surface

126 In 1966, Bluhm and Morrissey carried out an ambitious and pioneering study of the damage  
 127 evolution on cylindrical bars made of three ductile metals under tensile loading [3]. To do it, they  
 128 used a device (see Figure 9a) that could apply loading in a progressive manner, allowing a gradual  
 129 crack propagation, and ultrasonic and metallographic techniques for identifying internal damage. In

130 this study they already identified several phenomena that helped to understand better the load-strain  
 131 curve of a tensile test with this type of materials (see Figure 9b)):

- 132 • After necking localises, small microvoids develop, interfacial non-connected cracks at inclusions  
 133 appear, as well as short intercrystalline cracks.
- 134 • Later, as strain increases, the volume of voids and cracks also increase, weakening the material  
 135 matrix in the center of the necking region. Around this weakened zone, an essentially non-fractured  
 136 region remains under a low hydrostatic stress state, thus being under high shear stresses that  
 137 eventually lead to cracking out of the initial fracture plane, which are the so-called shear lips.

138 This explains the typical cup-cone fracture pattern, which would be deeply studied later by  
 139 several researchers.



**Figure 9.** a) Device used by Bluhm and Morrisey to study fracture on cylindrical bars made of different metals, b) Evolution of the damage process in the necking region of a copper cylindrical specimen under tension [3]

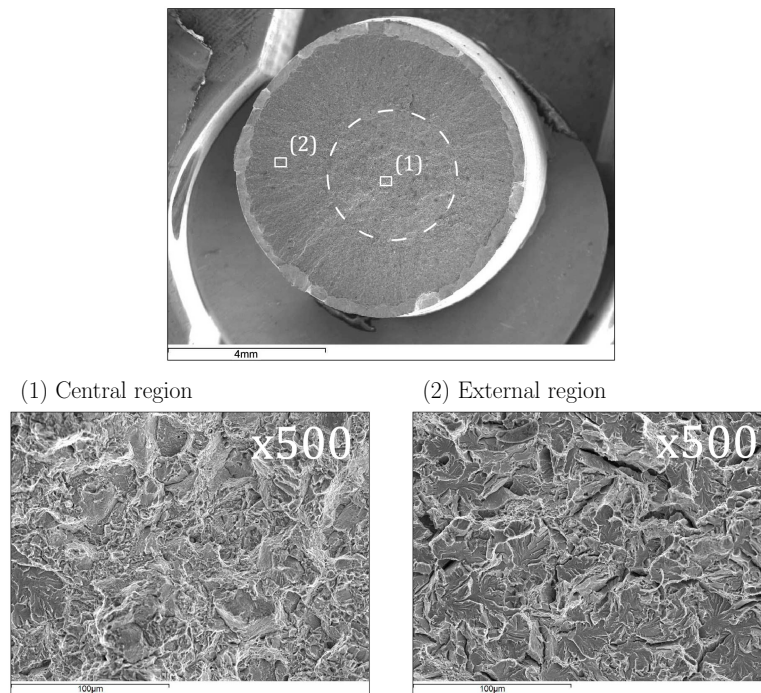
140 In this section the fracture surface on both analysed patterns, type 1 (flat fracture pattern) and  
 141 type 2 (cup-cone pattern), are studied by means of a scanning electronic microscope (SEM) and by  
 142 analysing the geometry of both fracture surfaces for specimens of distinct diameters.

### 143 2.3.1. Fractographs

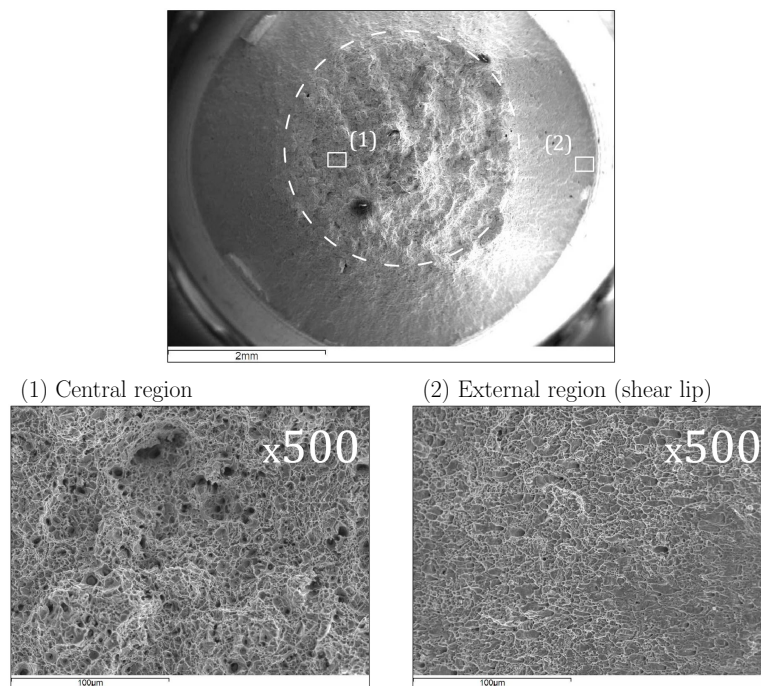
144 Figure 10 shows a 9 mm-diameter specimen of Material 1 observed with a SEM; the dashed  
 145 circumference represents the dark area that can be observed on the fracture surface with a mere visual  
 146 inspection (see Figure 1a)). The central region, which corresponds to the characteristic dark area  
 147 of this pattern, presents dimples, usually related with a nucleation-growth-coalescence mechanism,  
 148 while the surrounding region shows the so-called river marking related with a cleavage mechanism. In  
 149 Figure 11, the analogous SEM analysis of a 9 mm-diameter specimen of Material 2 can be observed; the  
 150 dashed circumference identifies the circular flat plane of the cup-cone pattern. These images show that  
 151 the central region is the result of a nucleation-growth-coalescence process while the surface observed  
 152 in the shear lips show a certain development of cleavage, probably mixed with nucleation-growth and  
 153 coalescence of microvoids.

### 154 2.3.2. Size of the internal damage

155 The dark region observed in the fracture pattern of type 1 can be seen as an internal notch induced  
 156 by a nucleation-growth-coalescence process developed during the loading process. Following this idea,  
 157 the material behaviour would be ductile although the final failure mechanism would be triggered by a



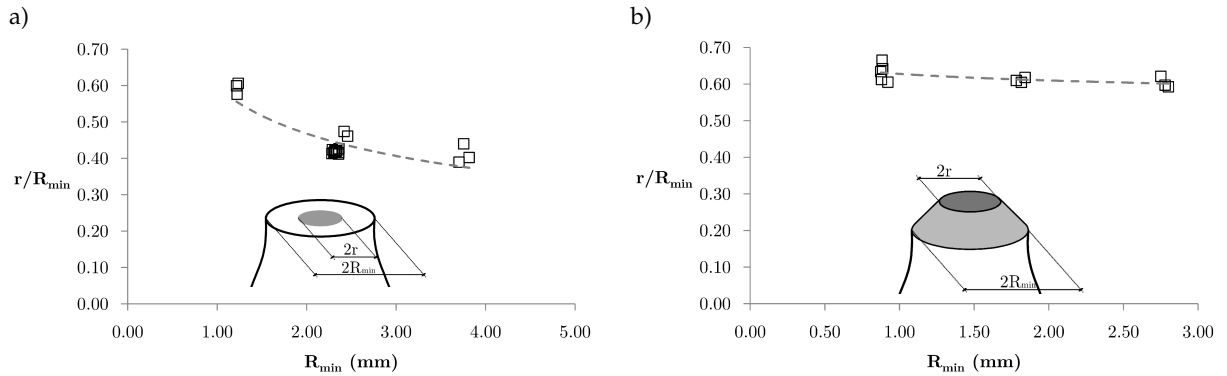
**Figure 10.** SEM analysis on the fracture surface of a 9 mm. diameter specimen of Material 1. [16]



**Figure 11.** SEM analysis on the fracture surface of a 9 mm. diameter specimen of Material 2. [16]

158 quasi-brittle behaviour, such quasi-brittle behaviour is explained by the presence of the dark region  
 159 which acts as an internal crack that, once large enough, concentrates stresses around a very specific  
 160 region of the damage cross-section and leads to a brittle fracture process at the very end of the test.

161 The area of the dark region observed in fracture surfaces of Material 1 does not seem to be  
 162 proportional to the specimen cross-section at first sight. If the dark region diameter ( $2r$ ) and the  
 163 minimum diameter of the specimen at necking ( $2R_{min}$ ) are measured, the evolution of  $r/R_{min}$  is  
 164 observed to decrease with the specimen size, as shown in Figure 12a). When the same analysis was  
 165 carried out on specimens of Material 2, considering  $2r$  as the diameter of the flat circular region of the  
 166 cup-cone surface, the proportion  $r/R_{min}$  remained constant for any specimen size (see Figure 12b)).



**Figure 12.**  $r/R_{min}$  relations experimentally obtained on specimens with diameters of 3, 6 and 9 mm for a) Material 1, b) Material 2 [23].

167 Under the aforementioned approach, the final fracture process can be approached by using  
 168 expressions based on LEFM for the case of Material 1, which are valid for brittle materials. Thus,  
 169 considering that the fracture process can be studied as the fracture of a cylindrical specimen with an  
 170 internal circular crack, the formula provided by Guinea, Rojo and Elices [24,25] can be used:

$$\frac{K_I}{K_0} = 1 + \sum_{i=1}^5 C_{i0} \left(\frac{r}{R}\right)^{\frac{2i+1}{2}} + \sum_{i=1}^3 \left\{ \ln \left[ 1 + \left(\frac{r}{R}\right)^{2i} \right] \cdot \left[ C_{i1} \ln^2 \left( \frac{b}{R} \frac{r}{R} \right) + \frac{C_{i2}}{\sqrt{\frac{b}{R} \frac{r}{R}}} \right] \right\} \quad (1)$$

where the reference stress intensity factor  $K_0$  is:

$$K_0 = \frac{2}{\pi} \cdot \sigma \sqrt{\pi r}$$

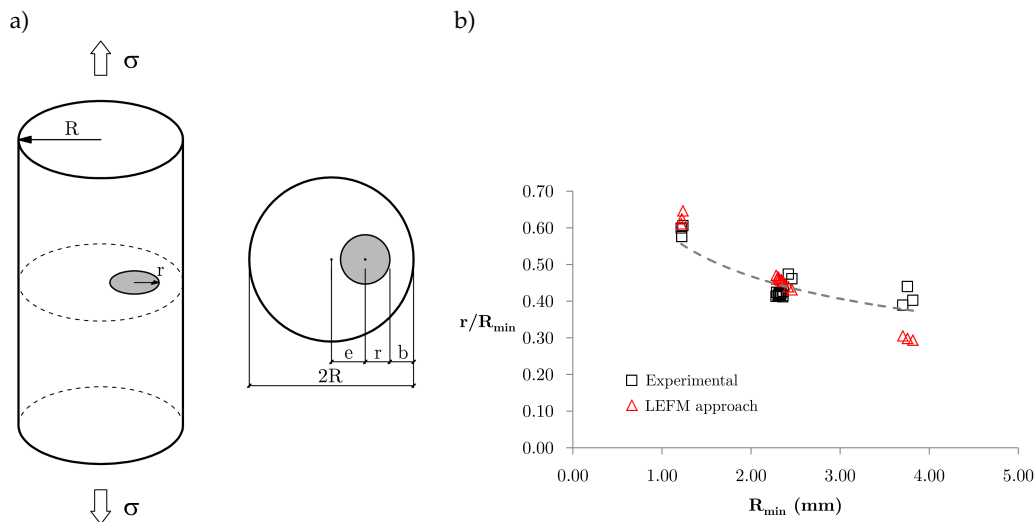
171 where (see Figure 13a)):

- 172 •  $K_I$  is the stress intensity factor.
- 173 •  $b$  stands for the smaller distance between the crack boundary and the specimen boundary.
- 174 •  $r$  is the internal crack radius.
- 175 •  $R$  is the specimen radius.
- 176 •  $\sigma$  is the tensile stress away from the fracture zone.
- 177 •  $C_i$  are the coefficients defined in Table 2.

**Table 2.**  $C_i$  coefficients used in the Guinea-Rojo-Elices expression for the computation of  $K_{Ic}$  in cylindrical fibers under tension with eccentric internal crack perpendicular to the specimen axis [24].

	$C_{i0}$	$C_{i1}$	$C_{i2}$
<b>i=1</b>	0,01242	-0,3097	1,185
<b>i=2</b>	-6,388	1,547	-3,723
<b>i=3</b>	16,89	-0,8769	2,628
<b>i=4</b>	-9,838	—	—
<b>i=5</b>	-1,228	—	—

178 The value of the fracture toughness  $K_{Ic}$  can be obtained by means of a three-point bending test,  
 179 described by the ASTM E 399-90 [26]. Therefore, once  $K_{Ic}$  is known, for each value of  $R_{min}$ , the value of  
 180  $r$  that satisfies (1) can be easily obtained. Figure 13 compares the values of proportion  $r/R_{min}$  obtained  
 181 experimentally and those obtained with (1), thus based on the LEFM. These results show that the  
 182 experimental values decrease with the specimen size, which agrees with the tendency predicted by  
 183 the LEFM, which supports the hypothesis of an eventual fracture driven by brittle mechanisms. In  
 184 contrast to these results, the almost constant  $r/R_{min}$  ratio exhibited by Material 2 specimens suggest a  
 185 final failure not ruled by LEFM considerations.



**Figure 13.** a) Parameters used in the expression of Guinea, Rojo and Elices [24], b)  $r/R_{min}$  values obtained with the expression based on the LEFM compared with the experimental results [23].

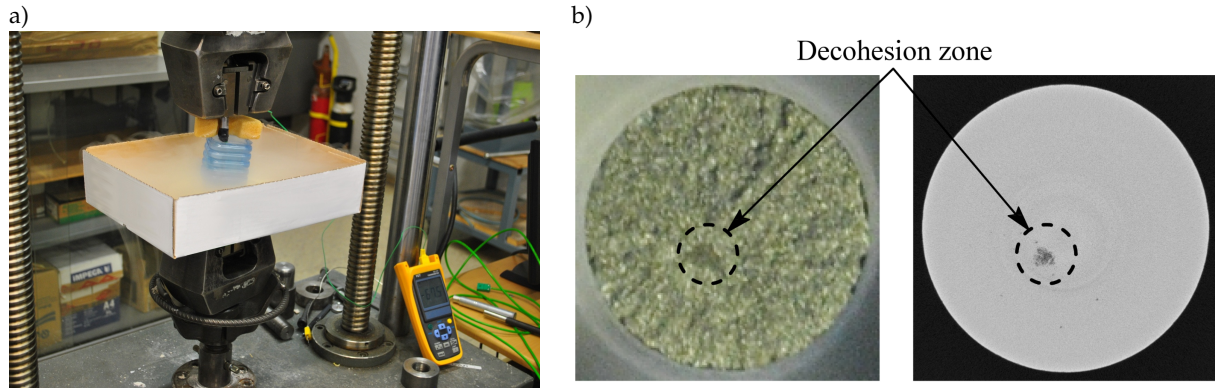
## 186 2.4. Evolution of damage

187 All the works mentioned before seem to confirm that, in the case of fracture pattern 1, damage  
 188 initiates inside the specimen producing something similar to an internal defect that eventually leads  
 189 to a brittle fracture process, but there is no evidence of when this damage begins to develop. In this  
 190 section two approaches are described which have helped to identify that the damage develops in the  
 191 very late stages of the tensile test.

### 192 2.4.1. Tests on embrittled specimens by means of liquid nitrogen

193 The stress-strain diagram of Material 1 is very repetitive, so a tensile test can be carried out to  
 194 a certain desired strain level with notable precision. In [15] four 6 mm specimens of this material  
 195 were tested up to increasing strain levels that ranged from the maximum load instant up to fracture,  
 196 with the highest of them being very close to failure. Each specimen was then unloaded and tested  
 197 again until failure at extremely low temperatures in order to induce an embrittled fracture that would

198 allow identification of the previously developed internal damage. Figure 14a) shows the experimental  
199 setup; the specimens were submerged into liquid nitrogen and the test was carried out with a rapid  
200 displacement of the clamping jaws once the temperature, measured with a thermo-couple, was lower  
201 than  $-100^{\circ}\text{C}$ .



**Figure 14.** a) Experimental setup for the test on specimens embrittled with liquid nitrogen, b) Comparison between the visual inspection after failure and the internal damage observed with X-ray computed tomography before failure [16].

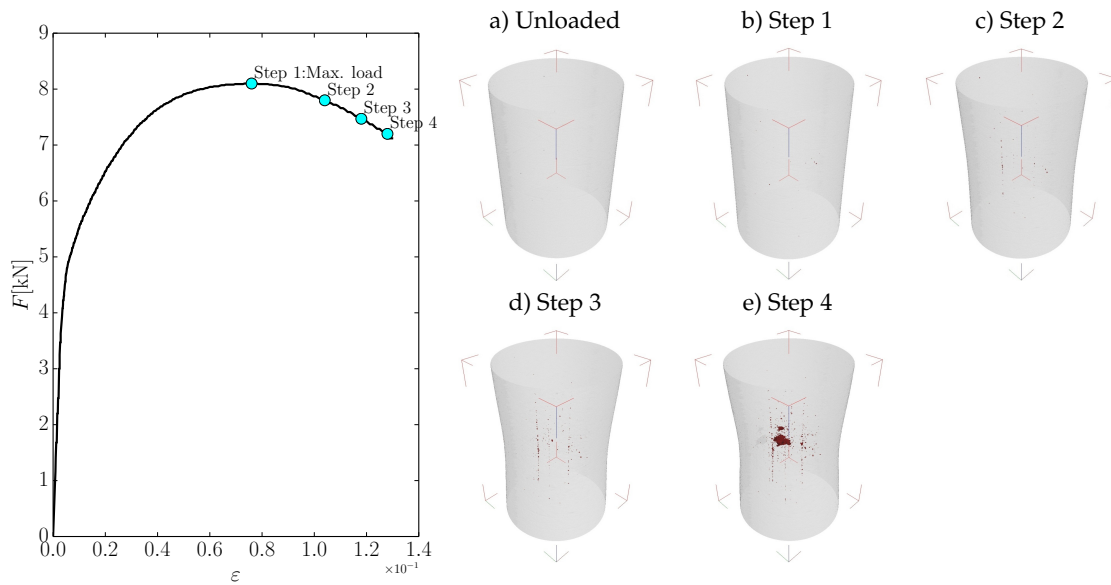
202 This procedure proved to be efficient and allowed to identify an internal damage only in the  
203 specimen tested at the more advanced strain level. In addition, this specimen was also analysed by  
204 X-ray computed tomography (XRCT) before the embrittled test. The tomographic image obtained also  
205 identified the internal damage that could be observed after carrying out the embrittled test. Figure  
206 14b) shows both results compared, which allows considering this region as the beginning of the dark  
207 region that can be usually observed in this fracture pattern, which can be considered as an internal  
208 decohesion process.

#### 209 2.4.2. Tests analysed with X-ray computed tomography

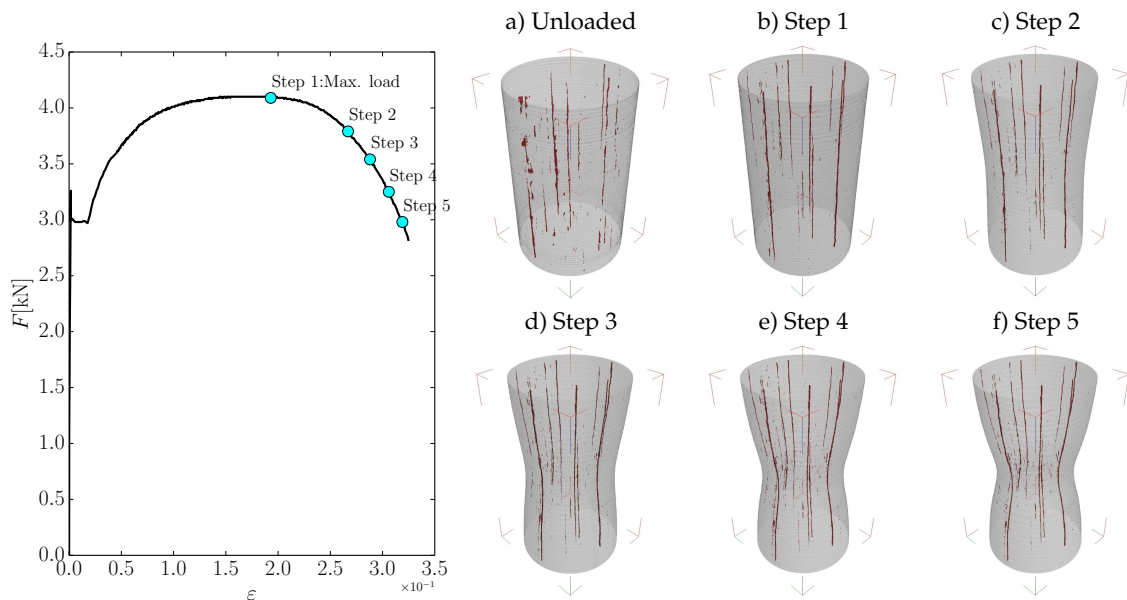
210 As seen before, XRCT allows identifying internal damage in the specimens, so a more detailed  
211 study of the damage evolution was done using this technique on both materials. For each of them, a 3  
212 mm diameter sample was tested up to a certain strain level, then unloaded and analysed with XRCT.  
213 Later on, the same sample was retested up to a higher strain level, unloaded and analysed with XRCT  
214 again. This process was repeated increasing the strain level until the sample was broken. Figures 15  
215 and 16 show the tomographic images obtained for both materials at each of the analysed strain levels.

216 In the case of Material 1, the matrix looks compact and homogeneous in the intact material  
217 and, as strain increases after maximum loading, the formation of small voids can be observed.  
218 Interestingly, some of these voids are formed following a longitudinal direction, which could be  
219 due to the manufacturing process. In this case, voids only coalesce at the very last stage of the test,  
220 corresponding to step 4 in the figure, in agreement with the behaviour observed in the 6 mm samples  
221 analysed with liquid nitrogen. Here, again, the decohesion process is identified, which will result in  
222 the dark region observed on the fracture surface of this material.

223 In the case of Material 2, the initial tomography, obtained with the intact specimen before testing,  
224 already shows a very heterogeneous matrix, with inclusions that are also lined up in the longitudinal  
225 direction, probably as a result of the manufacturing process. As strain increases, damage seems to  
226 develop all around the necking region.



**Figure 15.** Evolution of internal damage on a 3-mm diameter specimen of Material 1. Images obtained by means of X-ray computed tomography [16]



**Figure 16.** Evolution of internal damage on a 3-mm diameter specimen of Material 2. Images obtained by means of X-ray computed tomography [16]

## 2.5. Influence of stress triaxiality on ductile fracture

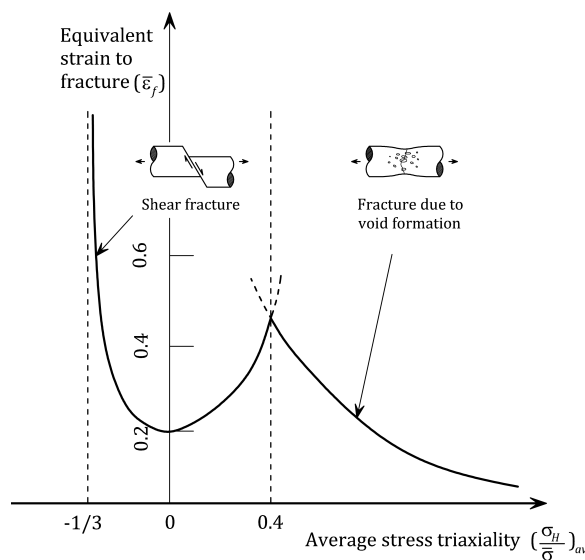
Stress triaxiality is defined as the ratio  $\sigma_H/\bar{\sigma}$ , with  $\sigma_H$  being the hydrostatic pressure and  $\bar{\sigma}$  the Von Mises equivalent stress. This value provides information of how balanced or unbalanced the principal stresses  $\sigma_1, \sigma_2$  and  $\sigma_3$  are.

Mirza et al. [27] studied the influence of the stress triaxiality using copper specimens; they performed tensile tests stopped before failure in order to, once unloaded, polish them to observe the matrix inside the specimen and identify internal damage. They concluded that the stress triaxiality plays a paramount role on how the nucleation-growth-coalescence mechanism develops. In this sense, Toribio et al. [28–30] also studied the influence of triaxiality using notched specimens to induce distinct triaxiality states in the fracture process.

The main material models developed until the mid-nineties included the stress triaxiality in their formulation in an implicit way, for example, by means of parameters  $p$  and  $q$  in the Gurson model [31] and by means of the Bridgman equations [32] in the Johnson-Cook [33] model.

The works by Bao and Wierzbicki [34–36] help to understand how this issue affects fracture on metallic specimens. They conducted a large experimental programme with alluminum specimens covering a wide range of triaxiality states, from compression to multiaxial tension. Figure 17 presents one of their most relevant results, a diagram that shows how the ultimate strain is dependent on the stress triaxiality. One of their main conclusions is that the influence of the stress triaxiality cannot be modelled by a monotonic function, as other authors had done in the past [33,37], but three distinct regions must be considered; in each region the governing fracture mechanisms are different:

- Zone I: low triaxialities, where fracture is mainly due to shearing.
- Zone II: medium triaxialities, where fracture is the result of a combination of shearing and the nucleation-growth-coalescence mechanism.
- Zone III: high triaxialities, where the nucleation-growth-coalescence mechanism drives fracture.



**Figure 17.** Influence of stress triaxiality on the equivalent strain to fracture (adapted from [35])

### 2.5.1. The Lode angle

In the last years, several models and approaches have been based on the so-called Lode angle [38]. If working with the principal stresses coordinate system  $(\sigma_1, \sigma_2, \sigma_3)$ , any stress state can be represented by means of the Haigh-Westergaard coordinates  $\xi, \rho$  and  $\theta$ :

$$\begin{aligned}\zeta &= \frac{I_1}{\sqrt{3}} = \sqrt{3}p \\ \rho &= \sqrt{2J_2} \\ \theta &= \frac{1}{3} \cos^{-1} \left( \frac{3\sqrt{3}}{2} \frac{J_3}{J_2^{3/2}} \right)\end{aligned}$$

255 where  $I_1$  is the first invariant of the Cauchy stress tensor,  $J_2$  and  $J_3$  stand for the invariants of the stress  
256 deviator tensor and  $p$  for the hydrostatic stress. Parameter  $\theta$  is usually referred to as the Lode angle.

257 In [39], a new approach to ductile fracture was proposed by using the Lode parameter, which in  
258 combination with the stress triaxiality is considered as fundamental to identify the fracture direction  
259 in ductile materials. In that work, Zhang et al. confirmed that the stress triaxiality is not enough to  
260 define the fracture behaviour of a ductile material and another value, in this case the Lode parameter,  
261 must be used. Since then, many authors have used the Lode parameter in ductile fracture models. For  
262 instance, Wue and Wierzbicki [40] proposed a model where the damage criterion was based on the  
263 stress triaxiality and a parameter related to the Lode angle, Bai and Wierzbicki [41] proposed later a  
264 ductile fracture model based on the stress triaxiality and the Lode parameter, and Erice and Gálvez  
265 [42,43] developed a coupled elastoplastic damage model with a failure criterion dependent on the  
266 Lode angle. The importance of the Lode angle in ductile fracture processes was confirmed by Mirone  
267 and Corallo [44], who studied its influence on fracture of different metals, and Barsoum and Faleskog  
268 [7,8,45], who based their work on the mechanisms of growth and coalescence of microvoids, concluded  
269 that the Lode parameter has a high influence on the nucleation and growth of voids but not much if  
270 the triaxiality is high, when it is the stress triaxiality which drives the fracture process.

### 271 3. Numerical models

#### 272 3.1. Models usually employed with metals

273 There are a number of material models that are currently used for simulating fracture in metals.  
274 The election of the model depends on the user's needs, for example, if the parameters that want to  
275 be considered in the fracture process involve strain rate, temperature or pressure, the Johnson-Cook  
276 model that will be briefly described later, would be a good choice.

277 If we consider the coupling of the elastoplastic behaviour and the fracture behaviour, we can  
278 divide these models in coupled and uncoupled. In the following lines, a brief overview of the main  
279 models used for reproducing fracture in metals is provided. Some of these formulations are the result  
280 of an analytical treatment of the problem and others are based on experimental adjustments of the  
281 problem. In any case, the reader will find that all of them include triaxiality as a key factor in the  
282 ductile fracture process and that some of them also include the Lode parameter.

283 After this overview, a model developed by the authors is described. This model is based on the  
284 cohesive crack concept and its formulation is affected by triaxiality, showing to reproduce reasonably  
285 well the fracture behaviour of Material 1.

#### 286 3.1.1. Uncoupled models

287 In these models, damage process does not affect the elastoplastic constitutive equations. A damage  
288 parameter must be defined, usually dependent on the plastic strain accumulation.

289 These models are usually phenomenological and, since the elastoplastic and failure criteria can be  
290 observed independently, are also usually easier to calibrate. Stress is not affected by a progressive  
291 deterioration process, which lead to rather an abrupt failure behaviour. Some fracture patterns, such  
292 as the cup-cone shape, cannot usually modelled by this type of formulations.

### 293 Johnson-Cook model

294 The Johnson-Cook model [33,46] is very extended when strain rate and thermal softening or  
 295 pressure are important factors to be considered in the fracture process, i.e. ballistic applications [47–51]  
 296 and blast loadings [52–55]. This model defines a damage parameter  $D$  expressed by (2), where  $\Delta\bar{\epsilon}_p$   
 297 is the equivalent plastic strain rate and  $\bar{\epsilon}_p^R$  the equivalent plastic strain to failure, that is dependent  
 298 on the strain rate, temperature and pressure, as expression (3) shows. In this expression  $D_1, D_2, D_3,$   
 299  $D_4$  and  $D_5$  are material constants that must be calibrated,  $\sigma^*$  stands for the stress triaxiality,  $\dot{\bar{\epsilon}}_p^*$  for a  
 300 dimensionless plastic strain rate ( $\dot{\bar{\epsilon}}_p^* = \dot{\bar{\epsilon}}_p / \dot{\epsilon}_0$ , with  $\dot{\epsilon}_0$  being a reference plastic strain rate) and  $T^*$  takes  
 301 into account the effect of temperature.

$$D = \sum \frac{\Delta\bar{\epsilon}_p}{\bar{\epsilon}_p^R} \quad (2)$$

$$\bar{\epsilon}_p^R = [D_1 + D_2 \exp(D_3\sigma^*)] [1 + D_4 \ln \dot{\bar{\epsilon}}_p^*] [1 + D_5 T^*] \quad (3)$$

### 302 Wilkins' et al. model

303 Wilkins' et al. material model [56] includes a scalar parameter  $A$  that plays a similar role as the  
 304 Lode parameter in later models and depends on the principal deviatoric stresses ( $s_1, s_2, s_3$ ) by means  
 305 of expression (4).

$$A = \max\left(\frac{s_2}{s_3}, \frac{s_2}{s_1}\right) \quad (4)$$

306 Then, the hardening function reads:

$$Y = Y_T(\bar{\epsilon}_p)A^\lambda + Y_S(\bar{\epsilon}_p)(1 - A^\lambda) \quad (5)$$

307 where  $Y_T$  is the equivalent strain hardening function for uniaxial tension/compression,  $Y_S$  the  
 308 equivalent strain hardening function for pure shear/torsion and  $\bar{\epsilon}_p$  the equivalent plastic strain.  
 309 The parameter  $\lambda$  is a material constant and must be experimentally adjusted.

310 The damage is dependent on the history of the plastic strain and its evolution depends on the  
 311 plastic strain rate and two separate weighting functions (see expression (6)), one depending on the  
 312 hydrostatic pressure ( $p$ ) and another depending on the parameter  $A$ .

$$\dot{D} = w_1(p)w_2(A)\dot{\bar{\epsilon}}_p \quad (6)$$

### 313 Bai-Wierzbicki model

314 The model proposed by Bai and Wierzbicki [41] explicitly introduces triaxiality dependence in the  
 315 hardening function, expressed by (7), where  $\bar{\sigma}(\bar{\epsilon}_p)$  represents the equivalent strain hardening function  
 316 for a reference test,  $\sigma_0^*$  the reference value of the stress triaxiality in the reference test,  $\bar{\epsilon}_p$  the equivalent  
 317 plastic strain,  $\sigma^*$  the stress triaxiality,  $c_{\sigma^*}, c_\theta^s$  and  $m$  are material constants and  $\gamma$  and  $c_\theta^{ax}$  parameters  
 318 dependent on the Lode angle  $\theta$ .

$$\phi = \bar{\sigma}(\bar{\epsilon}_p) [1 - c_{\sigma^*}(\sigma^* - \sigma_0^*)] \left[ c_\theta^s + (c_\theta^{ax} - c_\theta^s) \left( \gamma - \frac{\gamma^{m+1}}{m+1} \right) \right] \quad (7)$$

319 The damage accumulation is the same as in the Johnson-Cook model and thus depends on five  
 320 material constants that must be calibrated.

### 321 3.1.2. Coupled models

322 These models are usually based on micro-mechanical observations and the elastoplastic behaviour  
 323 is affected by the degradation process that leads to eventual failure. Because of their coupled nature,

324 their calibration is usually more tedious than in the case of uncoupled models, but are quite often  
325 selected by the scientific community due to their solid framework usually based on micromechanics.

### 326 *Lemaitre's model*

327 In this model [57,58], strain is obtained by expression (8), where  $D$  is a scalar variable that ranges  
328 from 0 to 1 and describes how damaged the material is,  $C$  is the elastic fourth-order tensor and  $(\varepsilon - \varepsilon_p)$   
329 the elastic strain.

$$\sigma = (1 - D)C : (\varepsilon - \varepsilon_p) \quad (8)$$

330 The yield function is defined by (9), where  $\bar{\sigma}$  is von Mises equivalent stress and  $\sigma_y$  is the isotropic  
331 hardening rule, dependent on a hardening variable  $r$ .

$$\phi = \frac{\bar{\sigma}}{(1 - D)} - \sigma_y(r) \quad (9)$$

332 Damage evolution is ruled by an expression that depends on the energy release rate, defined by  
333 (10).

$$Y = \frac{\bar{\sigma}^2}{2E(1 - D)^2} \left[ \frac{2}{3}(1 + \nu) + 3(1 - 2\nu)(\sigma^*)^2 \right] \quad (10)$$

334 where  $E$  and  $\nu$  are the material elastic parameters and  $\sigma^*$  the stress triaxiality.

### 335 *Xue-Wierzbicki model*

This model [59,60] couples damage and plasticity by means of expression (11), which resembles  
the expression (8), from Lemaitre's model.  $\beta$  is a material constant that must be calibrated using the  
matrix material stress-strain curve.

$$\sigma = (1 - D^\beta)C : (\varepsilon - \varepsilon_p) \quad (11)$$

336 The yield condition is defined by (12), where  $\sigma_M$  represents the yield strength of the matrix  
337 material and is a function of the plastic strain  $\varepsilon_p$ .

$$\phi = \bar{\sigma}^2 - \left[ (1 - D^\beta)\sigma_M \right]^2 \quad (12)$$

338 Damage evolves according to expression (13), where  $m$  is a material constant and  $\bar{\varepsilon}_p^f$  is the fracture  
339 strain from monotonic loading, which depends on the hydrostatic pressure  $p$  and the Lode angle  $\theta$ .

$$\dot{D} = m \left[ \frac{\bar{\varepsilon}_p}{\bar{\varepsilon}_p^f(p, \theta)} \right]^{(m-1)} \frac{1}{\bar{\varepsilon}_p^f(p, \theta)} \dot{\varepsilon}_p \quad (13)$$

### 340 *Modified Johnson-Cook model*

This model was proposed by Borvik et al. [61] and, based on the uncoupled Johnson-Cook model,  
reformulated as a coupled version. In this formulation, the strain rate is composed as the sum of three  
strain rates: elastic, plastic and thermally induced:

$$\dot{\varepsilon} = \dot{\varepsilon}^e + \dot{\varepsilon}^p + \dot{\varepsilon}^t$$

341 The stress tensor rate is defined by (14):

$$\dot{\sigma} = (1 - D)C : \dot{\varepsilon}^e - \frac{\dot{D}}{(1 - D)}\sigma \quad (14)$$

where  $C$  is the fourth-order elastic tensor and  $\dot{D}$  accounts for the damage evolution.

As in the original Johnson-Cook model, the damage evolution is defined as:

$$\dot{D} = \frac{\dot{\bar{\epsilon}}_p}{\bar{\epsilon}_p^R} \quad (15)$$

The proposed expression for the equivalent plastic strain to fracture  $\bar{\epsilon}_p^R$ , (16), is similar to that in the original Johnson-Cook model, although a different influence of the strain rate is considered.

$$\bar{\epsilon}_p^R = [D_1 + D_2 \exp(D_3 \sigma^*)] [1 + \dot{\bar{\epsilon}}_p^*]^{D_4} [1 + D_5 T^*] \quad (16)$$

### Gurson-like models

The Gurson-Tvergaard-Needleman (GTN) model is one of the most successful when fracture in metals is to be modelled. The seminal model was proposed by Gurson in 1977 [31], who based its formulation on the growth of a spherical void inside a material matrix, studied earlier by Rice and Tracey [62]. The model developed by Gurson could predict the loss of strength due to the nucleation and growth mechanisms, but not the total material loss of strength. Tvergaard and Needleman [4] modified the model including a failure criterion and were able to reproduce the typical cup-cone fracture. The yield function is defined by (17), where  $f^*$  represents the voids volume fraction of the material, which introduces the failure criterion based on the void volume at the beginning of the coalescence mechanism and the voids volume when there is a total loss of strength;  $q_1$ ,  $q_2$  and  $q_3$  are model parameters that, according to Tvergaard and Needleman, can be considered to be around 1.5, 1.0 and 2.25, respectively.

$$\phi = \left( \frac{\bar{\sigma}}{\sigma_y} \right)^2 + 2q_1 f^* \cosh \left( -q_2 \frac{3\sigma_H}{2\sigma_y} \right) - (1 + q_3 f^{*2}) \quad (17)$$

This model introduces the effect of triaxiality by means of the hydrostatic stress  $\sigma_H$  and the von Mises equivalent stress  $\bar{\sigma}$ . Since its appearance, it has been often used to study fracture in ductile materials and is still a referent model in the field. The model is able to reproduce ductile fracture in many materials [27,63,64], has been compared with other models proving to be very precise [65] and has even been used to study cases of ductile-brittle damage transition [66].

Many researchers have paid attention to this model and have adjusted its formulation to make it usable in other situations. For example, Hao and Brocks [67] proposed a variation that included the influence of temperature, Steglich et al. [68] combined the GTN and cohesive models to reproduce fracture in particle-reinforced metals, Zhang [39] modified the model formulation to reproduce coalescence in terms of the plastic limit by Thomason. In its original formulation, the GTN model is limited, since in cases of null hydrostatic pressure, damage does not develop; to solve this problem, Nahson and Hutchinson [69] proposed a modification that took into account cases of pure shear and cases where the hydrostatic pressure was very low; nevertheless, it was later proved that this version overestimated damage in cases with high triaxialities [70]. Later versions of the model by several authors can be found [64,71–74], some even including the Lode angle in their formulation [75,76].

In spite of the wide acceptance and usage of this model, it has also been found to present some drawbacks, mainly two: the parameters that feed the model cannot be experimentally measured, since they refer to voids volume fractions that cannot be quantified by experimental procedures, and distinct set of calibrated parameters can be found to be valid for the same material [68].

### 3.2. Triaxiality-dependent cohesive model

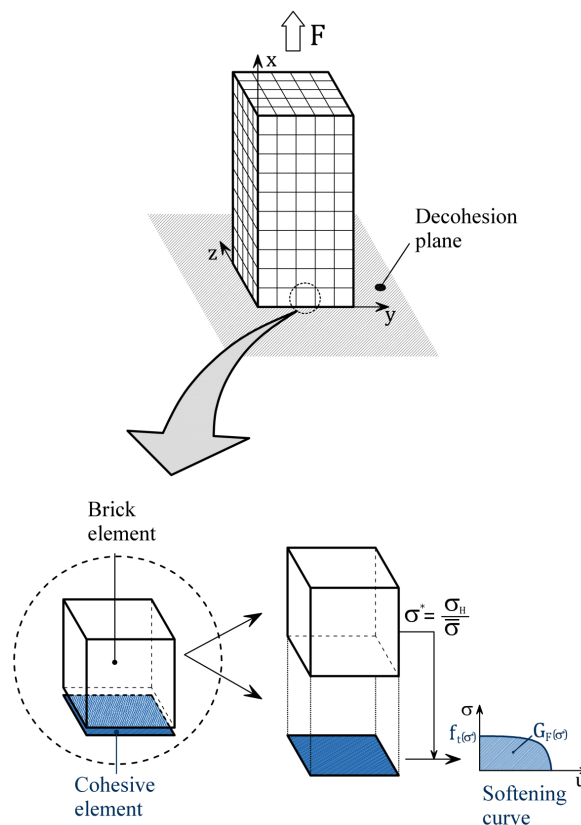
The cohesive crack concept was proposed in the seventies by Hillerborg [77], based on the work by Dugdale and Barenblatt [78], and was applied in the framework of Linear Elastic Fracture Mechanics (LEFM). Many models have appeared since then and have based the material damage on a softening

381 function that is defined by only two parameters, the tensile strength  $f_t$  and the fracture energy  $G_f$ . The  
 382 main advantages of these models are their simplicity and their easy calibration, which can be done by  
 383 standardised experimental tests [79,80], ensuring an objective procedure of calibrating and modelling  
 384 the fracture process. Besides this, they can be combined with any material model for the continuum,  
 385 including those discussed in the previous sections. For further details on this numerical model, the  
 386 reader is referred to [81] and [82].

387 This numerical approach has been used by many researchers since its appearance and has allowed  
 388 implementing interface cohesive elements for quasi-brittle materials under mode I or under the  
 389 combination of modes I, II and III loading [83–89], smeared crack models [90–94] and embedded crack  
 390 models [95–101].

391 Although the cohesive zone model is in principle valid in quasi-brittle materials, such as concrete  
 392 or cement-based materials, it has also been used in metals. Siegmund and Brocks [102] used a cohesive  
 393 crack model to study the crack growth in elastic-plastic materials and Scheider and Brocks [6,103] used  
 394 a cohesive model to simulate the cup-cone fracture and even made it dependent on triaxiality.

395 In [15], fracture pattern 2 was modelled by means of a cohesive zone model where the cohesive  
 396 parameters  $f_t$  and  $G_f$  were dependent on the stress triaxiality. To do this, an interface element  
 397 reproduced the fracture behaviour, which was fed by the stress triaxiality values of the adjacent  
 398 element (see Figure 18). In this case, since the fracture of a metal is reproduced, a rectangular-shaped  
 399 softening function is recommended [78]; given that for numerical reasons a rectangular shape induces  
 400 many convergence problems, a parabolic shape that resembles a rectangular behaviour is used.

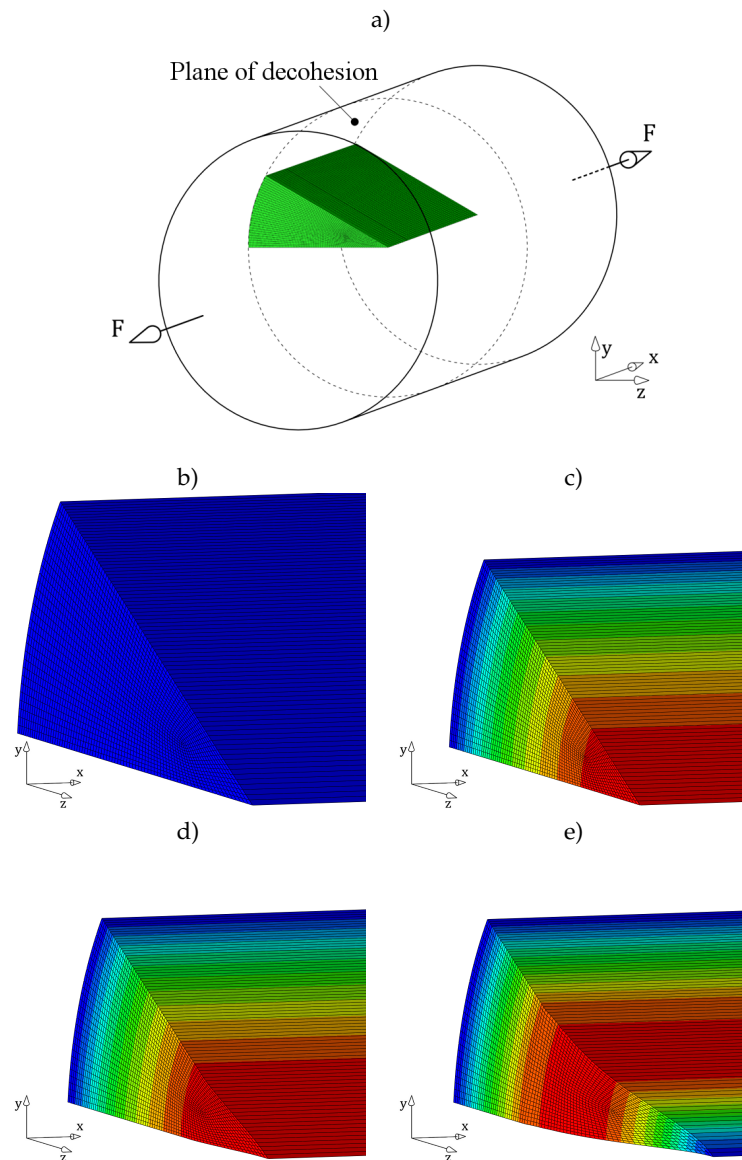


**Figure 18.** General graphical representation of the triaxiality-dependent cohesive zone interface element [15].

401 This model is fed by  $f_t$  and  $G_f$  values that can be obtained by standardised tests [26,104] and  
 402 reproduces reasonably well the decohesive process assumed in the center of the specimens that leads to  
 403 the dark region that can be observed in the fracture surface. As can be observed in Figure 19, when this

404 model is applied for modelling fracture in a cylindrical specimen of Material 1, fracture initiates in the  
 405 center of the specimen, unloading that part of the eventual fracture plane and, therefore, transferring a  
 406 stress increment to the surrounding region. As strain increases, this decohesive process propagates  
 407 from inside to outside.

408 This model provides reasonably good agreement with the experimental results, although it does  
 409 not provide a fracture criterion for the ductile-brittle transition observed in this type of fracture (see  
 410 Figure 1a)).



**Figure 19.** Evolution of decohesion at the fracture plane: a) grid used to simulate the fracture process, b) fracture plane before loading, c) stress concentrates in the center as loading is applied, d) decohesion begins, e) the decohesive process propagates from inside to outside. [15]

411 The similar approach has been successfully employed later [105] with gray cast iron, but instead  
 412 of by using interface elements, by means of crack-embedded elements.

#### 413 4. Final comments

414 In this paper, a review on the late advances made on the study of the two more usual fracture  
415 patterns observed on construction steels has been carried out, paying special attention to the material  
416 behaviour after maximum loading and the analysis of the fracture surfaces.

417 From the experimental point of view, although the ultimate strain values are usually neglected in  
418 practice, the experimental results carried out using a digital image correlation system, have proved to  
419 be very repetitive and independent from the specimen radius and length. Hence, in principle they  
420 could be used as reliable data if certain considerations are taken into account:

- 421 • If specimens of different radii are to be compared, a proportional-to-the-radius initial gauge  
422 length should be considered.
- 423
- 424 • The reference gauge length must be centered with the eventual fracture plane, otherwise the  
425 strain gradient would affect the measurements; this can be accomplished by using a digital image  
426 correlation system and not with conventional extensometers.

427 X-ray computed tomography allows identifying internal damage in steel, at least if thin enough  
428 specimens are used. This technique has helped to identify the internal damage evolution during a  
429 tensile test in the necking region. In the case of Material 1, which corresponds to an eutectoid steel used  
430 for manufacturing prestressing wires, the internal damage that eventually leads to a brittle fracture  
431 mechanism can be identified.

432 The study of the fracture surfaces geometries suggest that the fracture mechanisms of both  
433 analysed materials differ notably, not only for their final shape (cup-cone pattern and flat surface with  
434 internal dark circular region), but also because in the case of Material 1, this geometry proves not to  
435 be proportional to the specimen diameter, while in the case of Material 2, this proportionality can be  
436 observed.

437 A brief overview on the most extended numerical models for ductile materials has confirmed  
438 that the triaxiality dependence must always be considered in this type of fracture and, in many cases,  
439 also the influence of the Lode angle. The use of a cohesive crack formulation, has proved to provide  
440 reasonably good results even for metallic materials; this approach has been used by some researchers  
441 in the last years, including an application by the authors by using interface elements with the finite  
442 element method.

443 All the mentioned information can help to understand and support studying the behaviour of  
444 distinct steels beyond their maximum bearing capacity, which is usually neglected and considered  
445 unreliable. The works presented seem to point out in a different direction, since if the experimental  
446 work is carefully carried out, results seem to be pretty repetitive and reliable. A better understanding  
447 of this issue can help to extend the usage of these materials beyond their current limits, which is  
448 considered of interest, since the last part of the stress-strain diagram, between maximum loading  
449 and failure, is the main responsible for the energy absorbed by the material before collapsing. This  
450 could help to design better strategies in projected structures that could lead to higher structural safety  
451 conditions.

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456 Fernando Suárez and David A. Cendón carried out the numerical modelling; Fernando Suárez wrote the paper.

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